

Accounting for Variability in Design of Large-Scale Systems: A Reduced-Order Modeling Approach



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September 1, 2005
Sandia Workshop on Large-Scale Robust Optimization
Santa Fe, NM



Research supported by Air Force Research
Laboratory, Dr. Charles Cross

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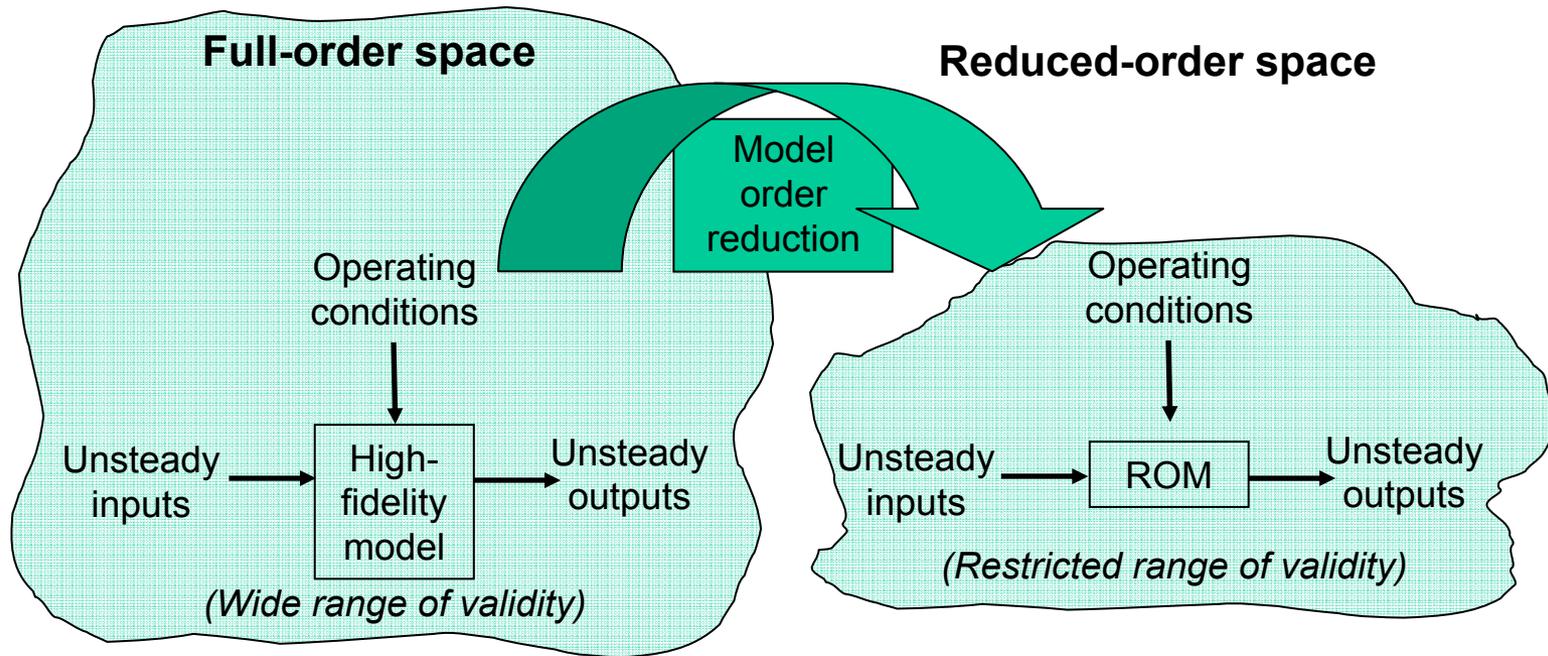
Outline

- Model reduction overview
 - Projection framework
 - Proper orthogonal decomposition (POD)
- Towards robust design and optimization (T. Bui-Thanh)
 - Including variability in reduced-order models
 - Example: design of bladed disks
- An optimization approach to model reduction (B. van Bloemen Waanders, B. Bader, O. Ghattas)



Model Order Reduction

Systematic model order reduction to replicate high-fidelity results over a restricted range of inputs.



→ Reduction level is several orders of magnitude

e.g. 2-D Euler : from $\sim 10^4$ states to $\sim 10^1$.



Why Model Reduction?

When is the high-fidelity dynamical system model too expensive?

- Multidisciplinary applications
 - Aeroelasticity (fluid/structure)
 - Flow control
- Real-time applications
- Design and optimization
- Probabilistic computations

Key idea: trade upfront computational work (offline phase) for a fast model (online phase).



Dynamical Systems



$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u}$$

$$\mathbf{y} = \mathbf{C}\mathbf{x}$$

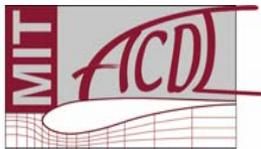
$$\dot{\mathbf{x}} = f(\mathbf{x}, \mathbf{u})$$

$$\mathbf{y} = g(\mathbf{x})$$

$\mathbf{x} \in \mathbf{R}^n$: state vector

$\mathbf{u} \in \mathbf{R}^p$: input vector

$\mathbf{y} \in \mathbf{R}^q$: output vector



CFD Dynamical Systems



$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u}$$

$$\mathbf{y} = \mathbf{C}\mathbf{x}$$

$$\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, \mathbf{u})$$

$$\mathbf{y} = \mathbf{g}(\mathbf{x})$$

- $\mathbf{x}(t)$: vector of n flow unknowns
e.g. 2D Euler, N grid points, $n = 4N$

$$\mathbf{x} = [\rho_1 (\rho u)_1 (\rho v)_1 e_1 \rho_2 \cdots \rho_N (\rho u)_N (\rho v)_N e_N]^T$$

- $\mathbf{u}(t)$: inputs
e.g. flow disturbances, wing motion
- $\mathbf{y}(t)$: outputs
e.g. flow characteristic, lift force



Reduced-Order Projection



$$\mathbf{x}(t) = \sum_{i=1}^m V_i \alpha_i(t)$$

$$\mathbf{x}(t) = V \mathbf{x}_r(t)$$

$$n \times 1 \left\{ \left[\begin{array}{c} \mathbf{x} \\ \mathbf{x} \end{array} \right] = \left[\begin{array}{c} V \\ V \end{array} \right] \left[\begin{array}{c} \mathbf{x}_r \\ \mathbf{x}_r \end{array} \right] \right\} m \times 1$$

$$W^T V = I$$



Reduced-Order Dynamical Systems



$$n \times 1 \left\{ \begin{array}{l} \left[\begin{array}{l} x \\ \end{array} \right] = \left[\begin{array}{l} V \\ \end{array} \right] \left[\begin{array}{l} x_r \\ \end{array} \right] \end{array} \right\} m \times 1 \quad W^T V = I$$

$$\begin{array}{l} \dot{x} = f(x, u) \\ y = g(x, u) \end{array}$$



$$\begin{array}{l} \dot{x}_r = W^T f(Vx_r, u) \\ y_r = W^T g(Vx_r, u) \end{array}$$

$$\begin{array}{l} \dot{x} = Ax + Bu \\ y = Cx \end{array}$$



$$\begin{array}{l} \dot{x}_r = A_r x_r + B_r u \\ y_r = C_r x_r \end{array}$$

$$A_r = W^T A V, \quad B_r = W^T B, \quad C_r = C V$$



Reduced-Order Basis

- Determine the projection $\mathbf{x} = V\mathbf{x}_r$
where V contains m basis vectors

$$V = [\mathbf{v}_1 \ \mathbf{v}_2 \ \cdots \ \mathbf{v}_m]$$

so that $m \ll n$ and system dynamics are captured accurately: $\mathbf{y}_r \approx \mathbf{y}$

- Most, but not all, reduction techniques use projection framework

Proper Orthogonal Decomposition



- Consider M snapshots $x_1, x_2, \dots, x_M \in \mathcal{R}^n$ (instantaneous state solutions)

- Construct kernel $K \in \mathcal{R}^{n \times n}$

$$K = \sum_{j=1}^M x_j x_j^T$$

$n \times n$
eigenvalue
problem

- $V = [v_1, v_2, v_3, \dots]$ are eigenvectors of K with $\lambda_1 \geq \lambda_2 \geq \lambda_3 \dots$
- If V_P contains the first P eigenvectors, then $Q_P = V_P V_P^T$ is the optimal projection in a least squares sense:

$$\min_{Q_P} \sum_{i=1}^M \|x_i - Q_P x_i\|_2^2 = \sum_{i=P+1}^N \lambda_i$$

- Note: optimality and error bound applies to the reconstruction of sampled data, not to the ROM.



Method of Snapshots



Sirovich: the eigenvectors of the kernel are linear combinations of the snapshots:

$$v_i = \sum_{j=1}^M \alpha_j^i x_j$$

The eigenvalue problem becomes:

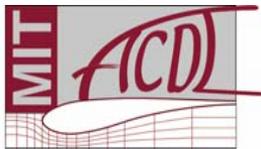
$$R\alpha^j = \lambda_j \alpha^j$$

$$\alpha^j = \begin{bmatrix} \alpha_1^j \\ \alpha_2^j \\ \vdots \end{bmatrix}$$

$M \times M$
eigenvalue
problem

where R is the $M \times M$ correlation matrix:

$$R_{ij} = x_i^T x_j$$





Time-Domain POD Algorithm

Reduction procedure:

1. Simulate the high-order CFD system to get M snapshots $\mathbf{x}_j, j=1,2,\dots,M$
2. Construct the correlation matrix $R_{ij} = \mathbf{x}_i^T \mathbf{x}_j$
3. Calculate the eigenvectors α^i and eigenvalues λ_i of R
4. Construct the basis functions

$$v_i = \sum_{j=1}^M \alpha_j^i x_j$$

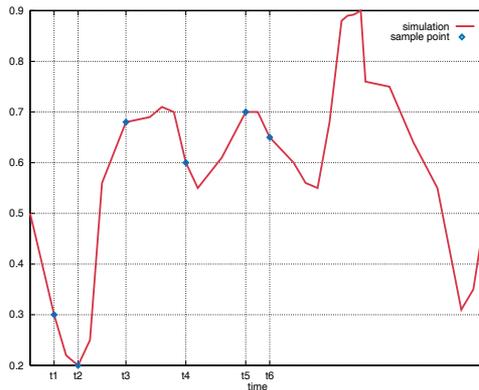
5. Select the most energetic basis functions using λ_i
6. Project the governing equations onto the reduced basis



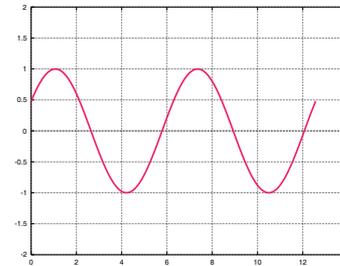
Most expensive step is the CFD simulation.

Frequency Domain

For linear systems, response can be decomposed into temporal harmonics:



$$= \Sigma$$



For $u = \bar{u}e^{j\omega t}$, $X = \bar{X}e^{i\omega t}$ and $y = \bar{y}e^{i\omega t}$

$$\bar{X} = [j\omega I - A]^{-1} B\bar{u}$$

$$\bar{y} = C\bar{X}$$

Frequency Domain POD

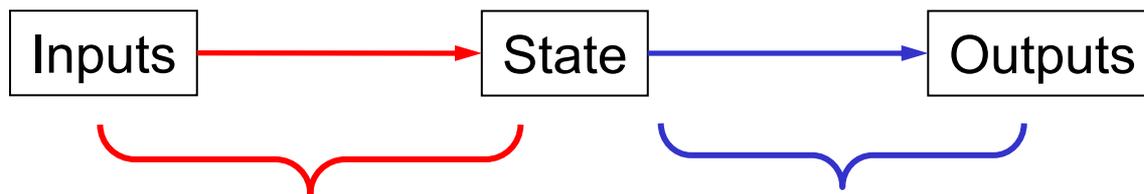
- Difficulty in choosing appropriate time simulation to obtain snapshots
- Instead, pick a set of sample frequencies ω_k
- Solve frequency domain equations at each frequency to obtain complex snapshots

$$\bar{X}(\omega_k) = [j\omega_k I - A]^{-1} B$$

$$K = \frac{1}{M} \sum_{k=1}^M \bar{X}(\omega_k) \bar{X}^*(\omega_k)$$

A Different View of POD

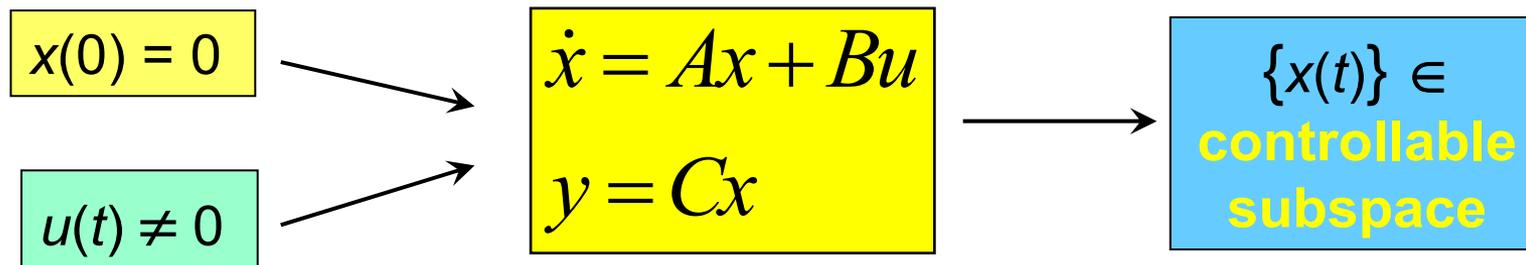
- POD often thought of in least squares context
 - POD basis vector provides least squares fit to snapshot data
- **Balanced truncation** provides different insight
 - Concepts of **controllability** and **observability**



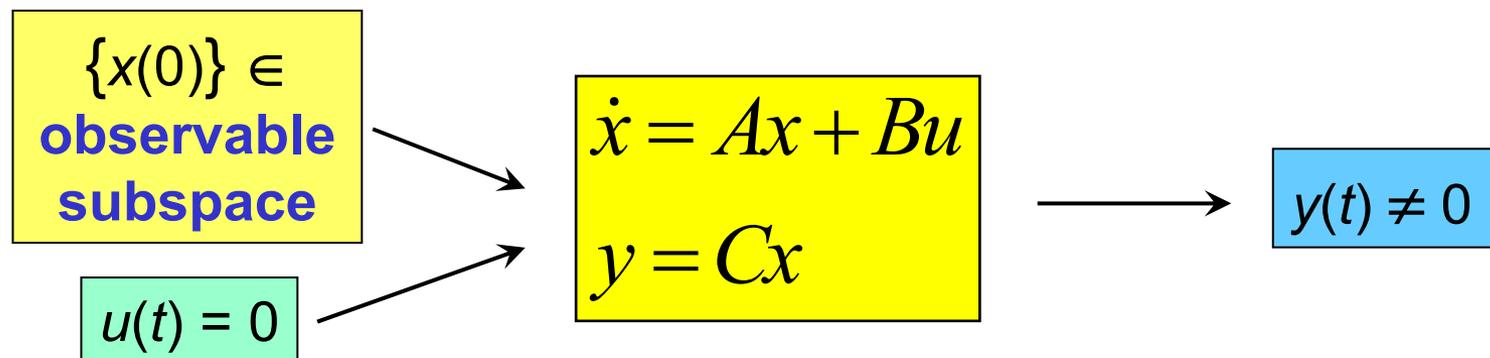
“Controllable” modes
– easy to reach, require small control energy

“Observable” modes
– generate large output energy

Subspaces



The **controllable subspace** is that set of states that can be obtained with zero initial state and a given input (reachable states).



The **observable subspace** is that set of states which as initial conditions produce a non-zero output with no external input.

Gramians

Controllability Gramian:

eigenvectors of W_c span the controllable subspace

$$W_c = \frac{1}{2\pi} \int_{-\infty}^{\infty} \underbrace{(j\omega I - A)^{-1} B B^T (-j\omega I - A^T)^{-1}}_{\bar{X}_\omega} d\omega$$

$(j\omega I - A)^{-1} B$ is the response to sinusoidal forcing

Observability Gramian:

eigenvectors of W_o span the observable subspace

$$W_o = \frac{1}{2\pi} \int_{-\infty}^{\infty} \underbrace{(j\omega I - A^T)^{-1} C^T C (-j\omega I - A)^{-1}}_{\bar{Z}_\omega} d\omega$$

$(j\omega I - A^T)^{-1} C^T$ is the dual response to sinusoidal forcing

POD Versus Balanced Truncation

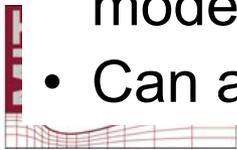
POD Kernel:

$$K = \frac{1}{M} \sum_{j=1}^M (j\omega_j I - A)^{-1} B B^* (-j\omega_j I - A^*)^{-1}$$

Controllability Gramian:

$$W_c = \frac{1}{2\pi} \int_{-\infty}^{\infty} (j\omega I - A)^{-1} B B^* (-j\omega I - A^*)^{-1} d\omega$$

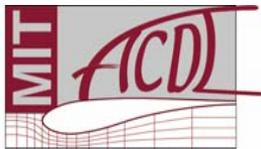
- POD is an approximation to the controllability Gramian with rectangular rule.
- POD basis vectors approximate the most controllable modes (over sampled frequencies).
- Can also include POD analysis of dual for observability.





Outline

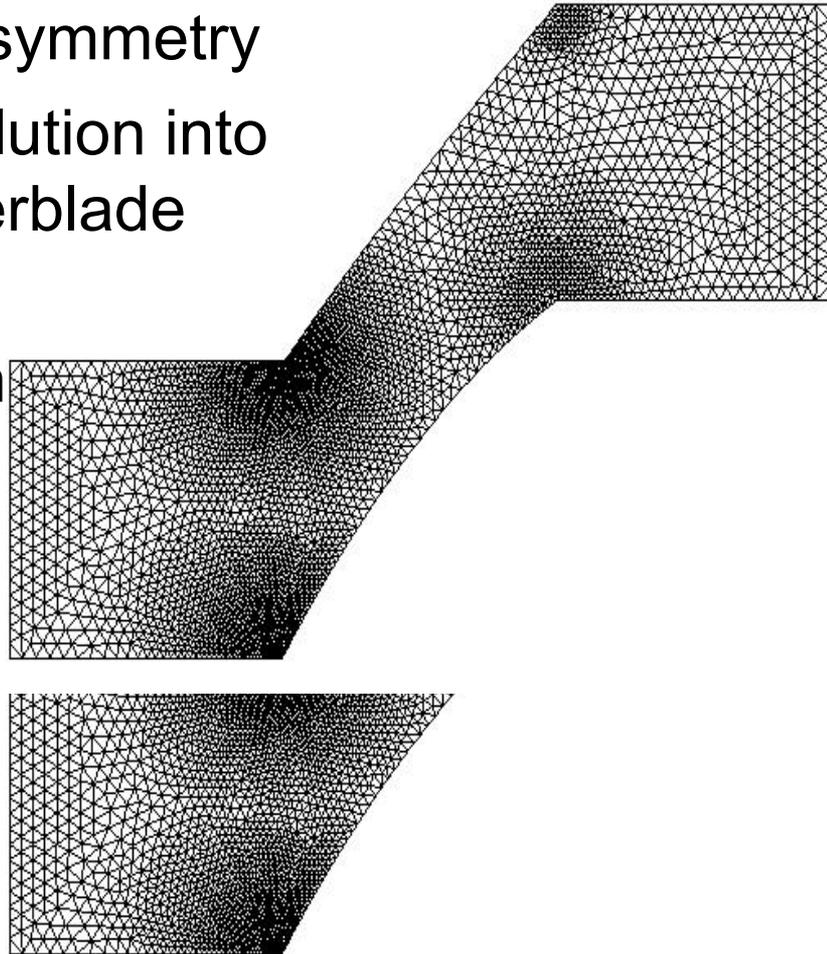
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- An optimization approach to model reduction



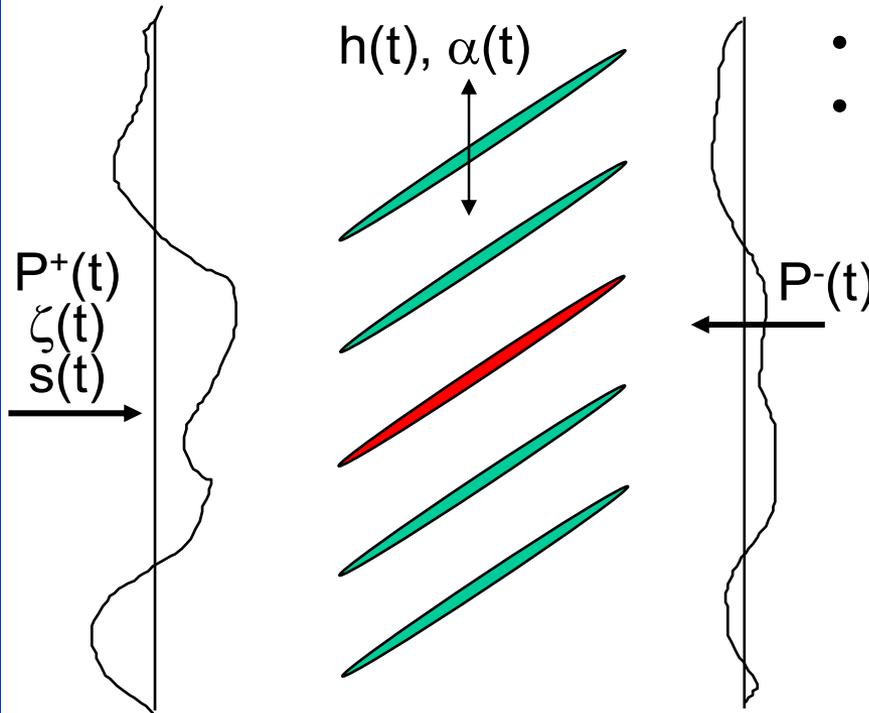
Turbomachinery Applications



- Problems have spatial symmetry
- Decompose general solution into spatial harmonics – interblade phase angles
- Solve linearized system efficiently in frequency domain on a single passage

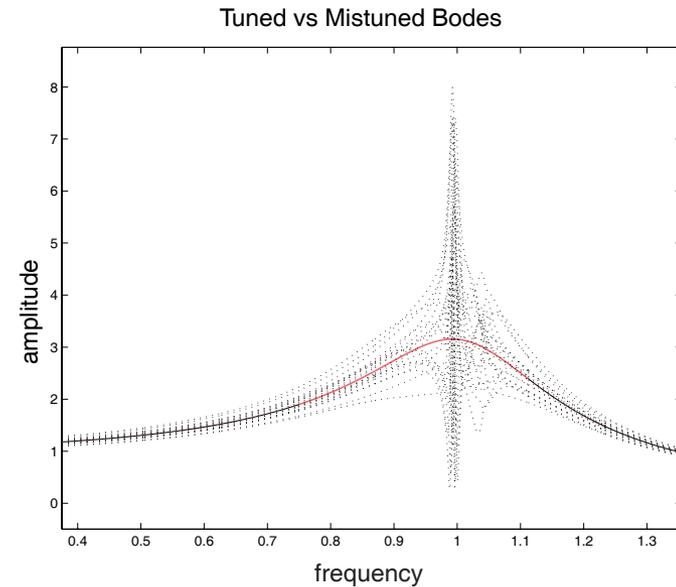


Effects of Variability

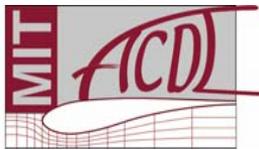


Variability in

- blade structural parameters
- blade shape



- Can have very large (negative) impact on forced response and a positive impact on stability
- Intentional mistuning can improve robustness
- Probabilistic design methods are a critical need



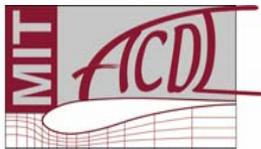
Representing Geometry Variability



- Mistuned blade geometry, g , can be expressed as (Garzon and Darmofal, 2003)

$$g = g_0 + \sum_{i=1}^{n_s} \sigma_i z_i w_i$$

- g_0 : nominal blade
- w_i : POD geometry modes
- $z_i \in N(0,1)$
- σ_i : standard deviation of the geometric data attributable to the i th mode
- n_s : number of mode shapes used to represent the mistuning.





Variability in CFD Models

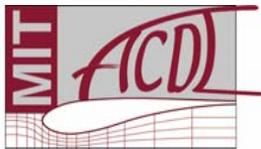
- CFD model dependence on geometry can now be written

$$\dot{\mathbf{x}} = A(\mathbf{z})\mathbf{x} + B(\mathbf{z})\mathbf{u}$$

$$\mathbf{y} = C(\mathbf{z})\mathbf{x}$$

where $\mathbf{z} = [z_1 \ z_2 \ \dots \ z_{n_s}]$.

- For probabilistic calculations, too expensive to evaluate A , B , C for many geometries



Approximate Model for Variability



- Efficient approximation using interpolation at nominal conditions, g_0 , and at selected interpolation points, z_i^j

$$A(z) \approx \bar{A}(z) = A_0 + \sum_{i=1}^{ns} \sum_{j=1}^{m_i} \frac{z_i \prod_{k=1, k \neq j}^m (z_i - z_i^k)}{z_i^j \prod_{k=1, k \neq j}^m (z_i^j - z_i^k)} (A_i^j - A_0)$$

- Interpretation:
 - Lagrange interpolation in each random variable z_i
 - Taylor series expansion
- Similar expressions for B , C



Approximate Model for Variability

- Approximate linearized system for arbitrary mistuning:

$$\dot{\mathbf{x}} = \left(A_0 + \sum_{i=1}^{ns} \sum_{j=1}^{m_i} \alpha_i^j (A_i^j - A_0) \right) \mathbf{x} + \left(B_0 + \sum_{i=1}^{ns} \sum_{j=1}^{m_i} \alpha_i^j (B_i^j - B_0) \right) \mathbf{u}$$

$$\mathbf{y} = \left(C_0 + \sum_{i=1}^{ns} \sum_{j=1}^{m_i} \alpha_i^j (C_i^j - C_0) \right) \mathbf{x}$$

$$\text{where } \alpha_i^j(z_i) = \frac{z_i \prod_{k=1, k \neq j}^m (z_i - z_i^k)}{z_i^j \prod_{k=1, k \neq j}^m (z_i^j - z_i^k)}$$

- Requires $\sum_{i=1}^{ns} m_i$ expensive linearization system evaluations (offline cost)
- Online cost to compute system for any mistuning \mathbf{z} is small.



Probabilistic Reduced-Order Model

- Given a reduced basis for the flow state, V, W :

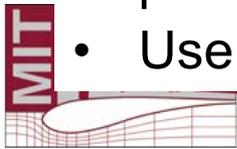
$$\dot{\mathbf{x}}_r = \left(\mathcal{A}_0 + \sum_{i=1}^{ns} \sum_{j=1}^{m_i} \alpha_i^j (\mathcal{A}_i^j - \mathcal{A}_0) \right) \mathbf{x}_r$$

$$+ \left(\mathcal{B}_0 + \sum_{i=1}^{ns} \sum_{j=1}^{m_i} \alpha_i^j (\mathcal{B}_i^j - \mathcal{B}_0) \right) \mathbf{u}$$

$$\mathbf{y}_r = \left(\mathcal{C}_0 + \sum_{i=1}^{ns} \sum_{j=1}^{m_i} \alpha_i^j (\mathcal{C}_i^j - \mathcal{C}_0) \right) \mathbf{x}_r$$

where $\mathcal{A} = W^T A V$, $\mathcal{B} = W^T B$, $\mathcal{C} = C V$

- Basis must capture both unsteady dynamics (i.e. t) and parameter variations (i.e. \mathbf{z})
- Use sampling methods proposed by Patera et al.



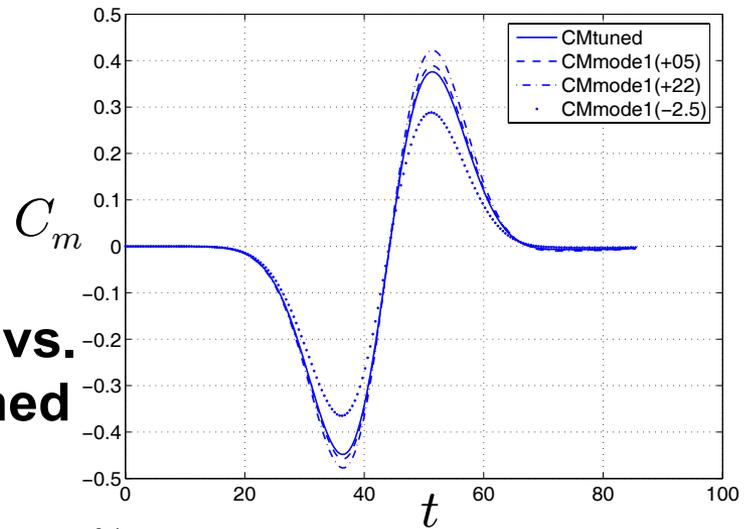
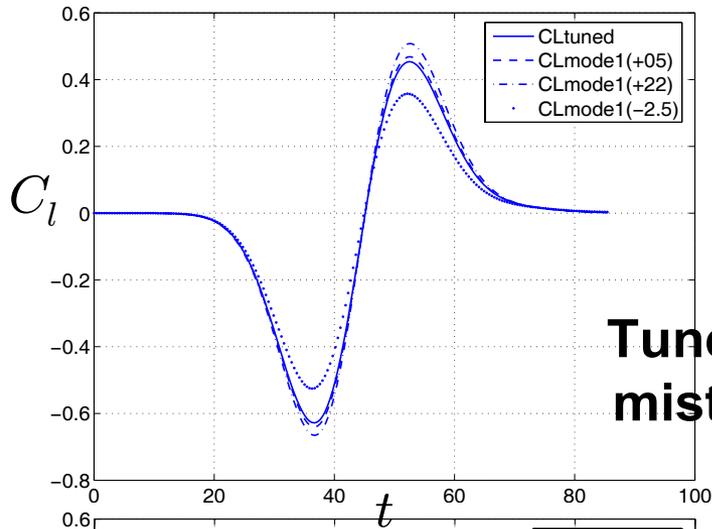


Example: Blade Shape Variability

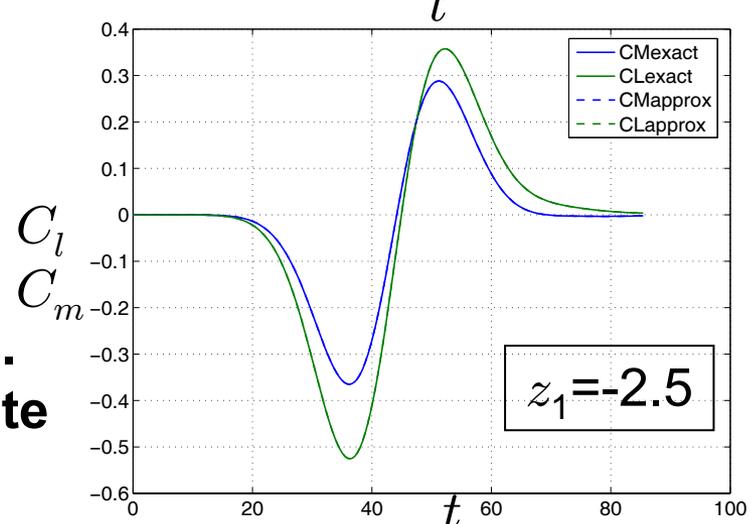
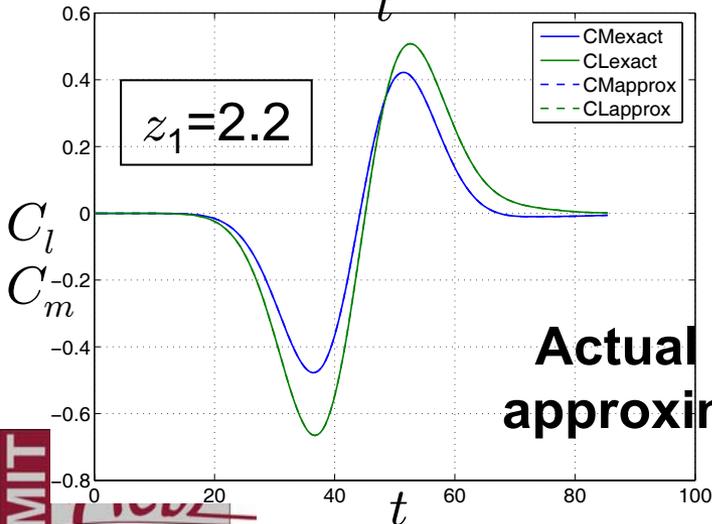
- Actual manufacturing data from Pratt & Whitney
 - 145 blades measured at 9 sections
- Preliminary analysis:
 - Blade in isolation
 - Two-dimensional analysis at mid-section
 - Rigid unsteady plunging motion
 - $M=0.3$
- Geometry model with 2 POD shape modes captures 82.55% of variability



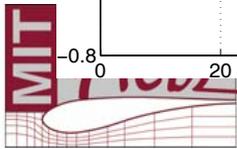
Geometry Mode 1



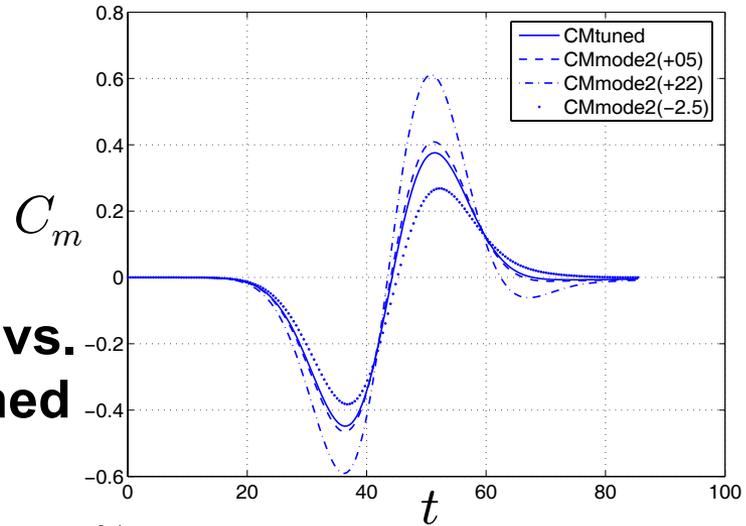
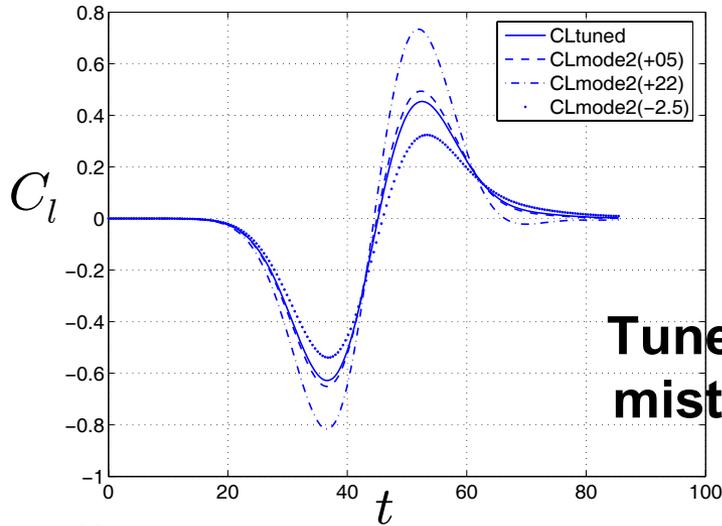
**Tuned vs.
mistuned**



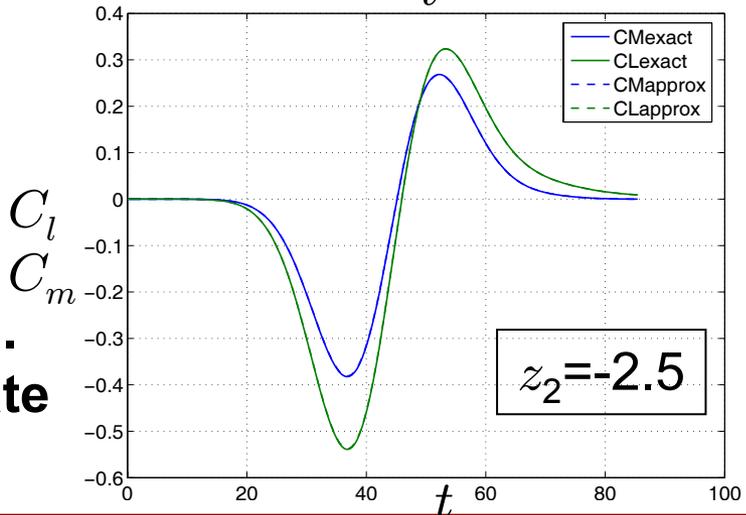
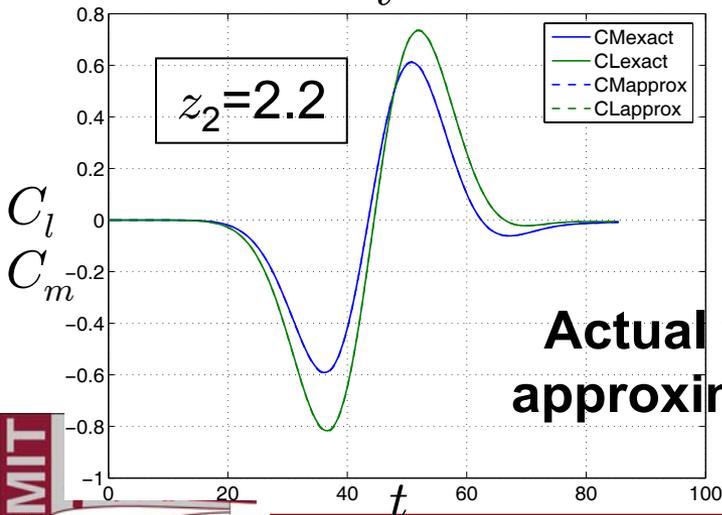
**Actual vs.
approximate**



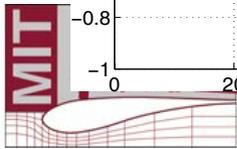
Geometry Mode 2



**Tuned vs.
mistuned**



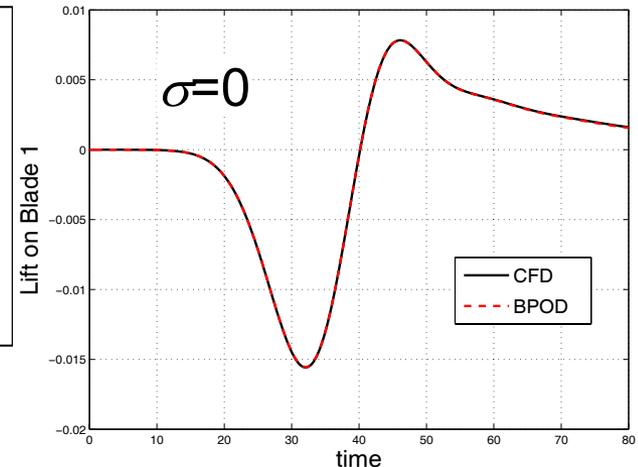
**Actual vs.
approximate**



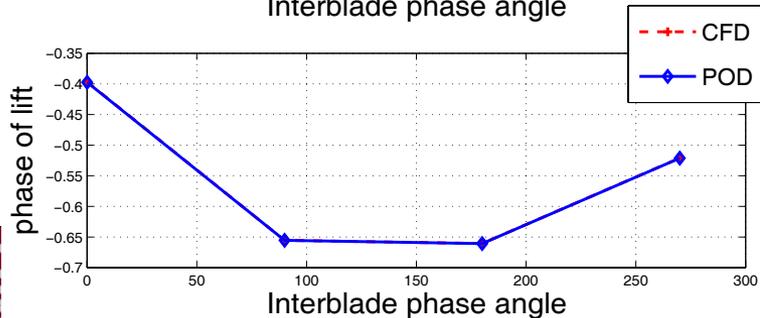
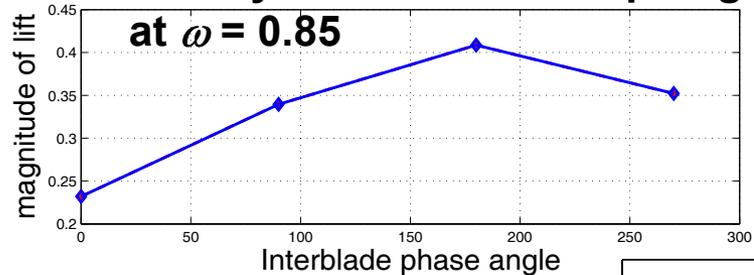
Tuned System ROM Results



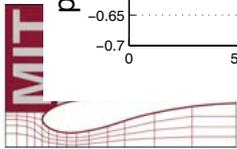
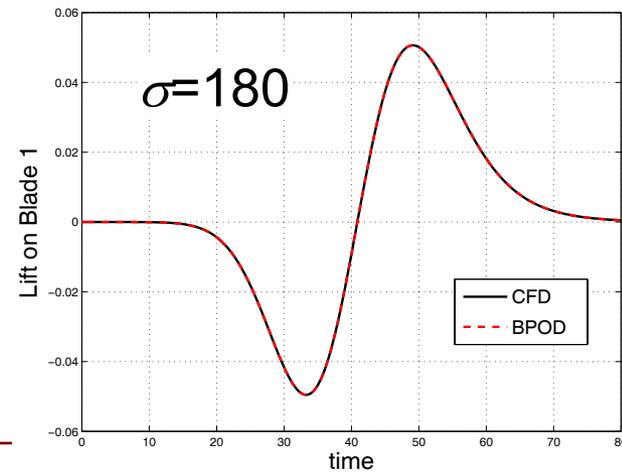
First standard configuration, 4 blades
CFD: 302,000 states
76,000 states per passage
ROM: 92 states
23 states for each IBP



Unsteady lift for harmonic plunge



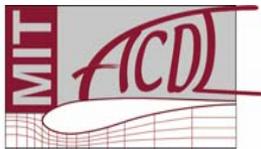
Unsteady lift for pulse input in plunge





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Reduction via POD

- POD is a data-driven approach. Basis and ROM do not respect the governing equations.
- Reduction via POD offers no guarantees of ROM quality (accuracy/stability).
- Proposed method: pose model reduction problem as an optimization problem.



Reduction via Optimization



$$\min_{V, \mathbf{x}_r} \mathcal{G} = \frac{1}{2} \sum_{k=1}^S \int_0^{t_f} (\mathbf{y}^k - \mathbf{y}_r^k)^T (\mathbf{y}^k - \mathbf{y}_r^k) dt$$
$$+ \frac{\beta}{2} \sum_{j=1}^m (1 - \mathbf{V}_j^T \mathbf{V}_j)^2$$

subject to

$$V^T V \dot{\mathbf{x}}_r^k = V^T A^k V \mathbf{x}_r^k + V^T B^k \mathbf{u}^k, \quad k = 1, \dots, S$$

$$V \mathbf{x}_r^k(0) = \mathbf{x}^k(0), \quad k = 1, \dots, S$$

$$\mathbf{y}_r^k = C^k V \mathbf{x}_r^k, \quad k = 1, \dots, S$$



Optimization Framework: Goal-Oriented



$$\min_{V, \mathbf{x}_r} \mathcal{G} = \frac{1}{2} \sum_{k=1}^S \int_0^{t_f} (\mathbf{y}^k - \mathbf{y}_r^k)^T (\mathbf{y}^k - \mathbf{y}_r^k) dt$$

Error in reduced-order
outputs for sample set

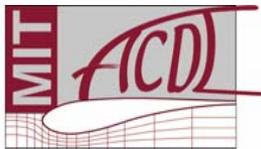
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Optimization Framework



$$\min_{V, \mathbf{x}_r} \mathcal{G} = \frac{1}{2} \sum_{k=1}^S \int_0^{t_f} (\mathbf{y}^k - \mathbf{y}_r^k)^T (\mathbf{y}^k - \mathbf{y}_r^k) dt$$

$$+ \frac{\beta}{2} \sum_{j=1}^m (1 - \mathbf{V}_j^T \mathbf{V}_j)^2$$

subject to

$$V^T V \dot{\mathbf{x}}_r^k = V^T A^k V \mathbf{x}_r^k + V^T \mathbf{b}^k, \quad k = 1, \dots, S$$

$$V \mathbf{x}_r^k(0) = \mathbf{x}^k(0), \quad k = 1, \dots, S$$

$$\mathbf{y}_r^k = C^k V \mathbf{x}_r^k, \quad k = 1, \dots, S$$

Regularization term to yield basis vectors of unit length



Optimization Framework: Model-Based



$$\min_{V, \mathbf{x}_r} \mathcal{G} = \frac{1}{2} \sum_{k=1}^S \int_0^{t_f} (\mathbf{y}^k - \mathbf{y}_r^k)^T (\mathbf{y}^k - \mathbf{y}_r^k) dt$$
$$+ \frac{\beta}{2} \sum_{j=1}^m (1 - \mathbf{V}_j^T \mathbf{V}_j)^2$$

subject to

Reduced output predictions from solution of governing equations

$$V^T V \dot{\mathbf{x}}_r^k = V^T A^k V \mathbf{x}_r^k + V^T B^k \mathbf{u}^k, \quad k = 1, \dots, S$$
$$V \mathbf{x}_r^k(0) = \mathbf{x}^k(0), \quad k = 1, \dots, S$$
$$\mathbf{y}_r^k = C^k V \mathbf{x}_r^k, \quad k = 1, \dots, S$$



Basis Computation

- Time integrals in objective function are replaced by a summation over a finite number of discrete time instants
 - Method requires a priori computation of a snapshot set over S parameter instances and T timesteps
- Assume the basis vectors are linear combinations of snapshots

$$\mathbf{v}_j = \sum_{i=1}^{ST} \gamma_i^j \mathbf{x}_i \quad j = 1, \dots, m$$

- Reduces number of unknowns from mn to mST
 - Typically $ST \ll n$
- Use POD basis as initial guess for optimizer

Optimization Framework vs. POD

$$\min_{V, \mathbf{x}_r} \mathcal{G} = \frac{1}{2} \sum_{k=1}^S \int_0^{t_f} (\mathbf{x}^k - \hat{\mathbf{x}}^k)^T C^{kT} C^k (\mathbf{x}^k - \hat{\mathbf{x}}^k) dt$$

$$+ \frac{\beta}{2} \sum_{j=1}^m (1 - \mathbf{V}_j^T \mathbf{V}_j)^2$$

subject to

$$V^T V \dot{\mathbf{x}}_r^k = V^T A^k V \mathbf{x}_r^k + V^T B^k \mathbf{u}^k, \quad k = 1, \dots, S$$

$$V \mathbf{x}_r^k(0) = \mathbf{x}^k(0), \quad k = 1, \dots, S$$

$$\hat{\mathbf{x}}^k = V \mathbf{x}_r^k$$

**OPT: min error
in *computed*
data**

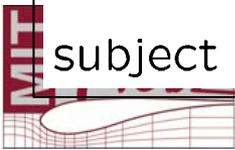
$$\min_V \mathcal{G}_{\text{pod}} = \frac{1}{2} \sum_{k=1}^S \int_0^{t_f} (\mathbf{x}^k - \tilde{\mathbf{x}}^k)^T (\mathbf{x}^k - \tilde{\mathbf{x}}^k) dt$$

$$+ \frac{\beta}{2} \sum_{j=1}^m (1 - \mathbf{V}_j^T \mathbf{V}_j)^2$$

subject to

$$\tilde{\mathbf{x}}^k(t_j) = V V^T \mathbf{x}^k(t_j)$$

**POD: min error
in *projected*
data**



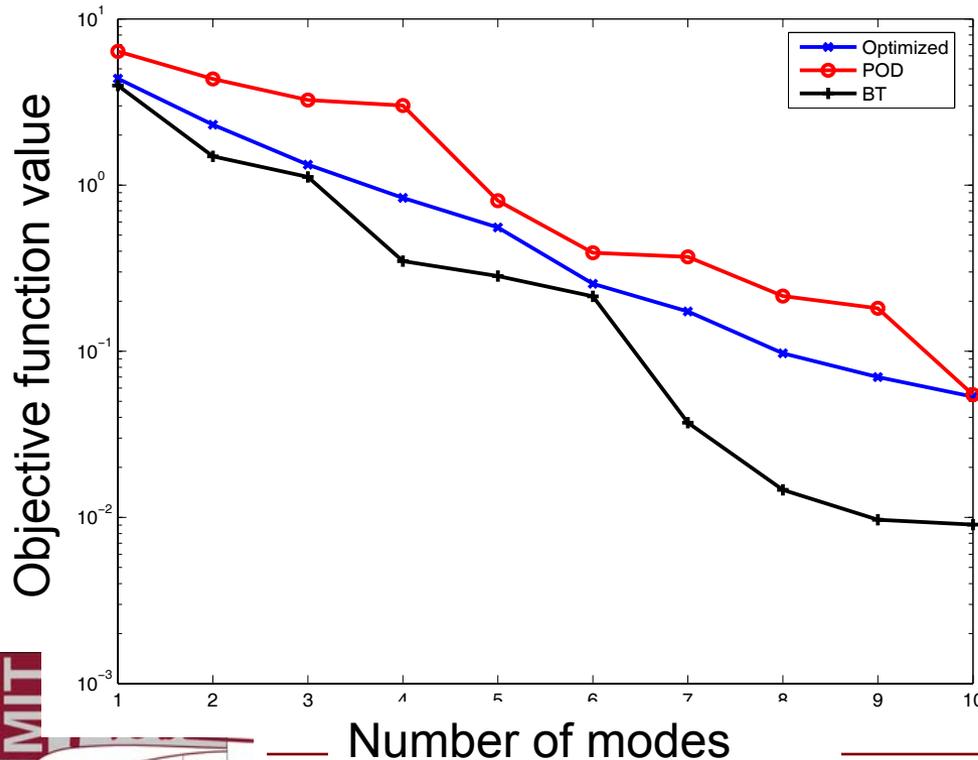
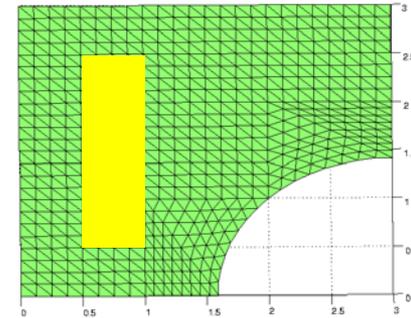
2D Heat Conduction Example



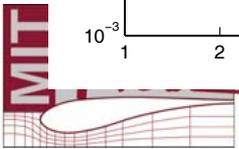
$$\frac{\partial \bar{u}}{\partial t} - k \nabla^2 \bar{u} = 0 \text{ in } \Omega$$

$$\bar{u} = \bar{u}_c \text{ on } \Gamma$$

$$\bar{u} = \bar{u}_0 \text{ in } \Omega \text{ for } t = 0$$



$S=5$ parameter instances
 $T=20$ time steps
 $n=480$ states
 $q=47$ outputs





Summary

- Model reduction methodology provides an efficient representation both for unsteady dynamics and for variability
- Formulation leads to parametric reduced-order models
 - Challenge: sampling methods to create a basis that spans high-dimensional input space
- Model reduction problem can be solved using optimization
 - An opportunity for robust optimization?
- Remaining challenge: incorporation of reduced-order models into probabilistic design and optimization

